

AD-A138 911

PULSED D2-F2 CHAIN-LASER DAMAGE TO COATED WINDOW AND  
MIRROR COMPONENTS..(U) AEROSPACE CORP EL SEGUNDO CA  
AEROPHYSICS LAB S T AMIMOTO ET AL. 15 DEC 83

1/1

UNCLASSIFIED

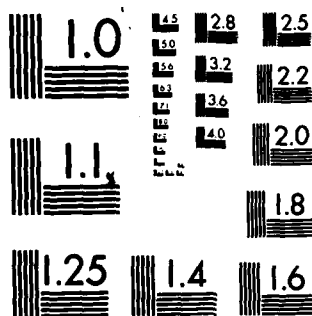
TR-0084(4930-01)-2 SD-TR-83-92

F/G 20/5

NL



END  
DATE  
FILMED  
4-84  
DTIC



MICROCOPY RESOLUTION TEST CHART  
NATIONAL BUREAU OF STANDARDS-1963-A

(12)

AD A1 38911

# Pulsed D<sub>2</sub>-F<sub>2</sub> Chain-Laser Damage to Coated Window and Mirror Components

S. T. AMIMOTO, J. S. WHITTIER, A. WHITTAKER,  
A. CHASE, and R. HOFLAND, Jr.  
Aerophysics Laboratory  
Laboratory Operations  
The Aerospace Corporation  
El Segundo, Calif. 90245

and

M. BASS  
USC Center for Laser Studies  
Los Angeles, Calif. 90009

15 December 1983

APPROVED FOR PUBLIC RELEASE;  
DISTRIBUTION UNLIMITED

DTIC  
ELECTE  
MAR 13 1984  
A

DTIC FILE COPY

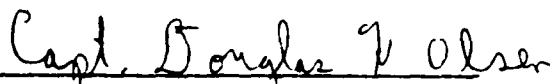
Prepared for  
SPACE DIVISION  
AIR FORCE SYSTEMS COMMAND  
Los Angeles Air Force Station  
P.O. Box 92960, Worldway Postal Center  
Los Angeles, California 90009

84 03 13 009

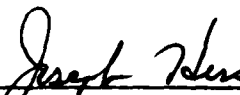
This report was submitted by The Aerospace Corporation, El Segundo, CA 90245, under Contract No. F04701-83-C-0084 with the Space Division, P.O. Box 92960, Worldway Postal Center, Los Angeles, CA 90009. It was reviewed and approved for The Aerospace Corporation by W. P. Thompson, Director, Aerophysics Laboratory. Captain D. H. Olsen, SD/YNS, was the project officer for the Mission Oriented Investigation and Experimentation (MOIE) Program.

This report has been reviewed by the Public Affairs Office (PAS) and is releasable to the National Technical Information Service (NTIS). At NTIS, it will be available to the general public, including foreign nationals.

This technical report has been reviewed and is approved for publication. Publication of this report does not constitute Air Force approval of the report's findings or conclusions. It is published only for the exchange and stimulation of ideas.



Douglas H. Olsen, Captain, USAF  
Project Officer



Joseph Hess, GM-15, Director  
West Coast Office, Air Force Space  
Technology Center

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

REPORT DOCUMENTATION PAGE		READ INSTRUCTIONS BEFORE COMPLETING FORM								
1. REPORT NUMBER SD-TR-83-92	2. GOVT ACCESSION NO. <b>A138911</b>	3. RECIPIENT'S CATALOG NUMBER								
4. TITLE (and Subtitle) PULSED D <sub>2</sub> -F <sub>2</sub> CHAIN-LASER DAMAGE TO COATED WINDOW AND MIRROR COMPONENTS		5. TYPE OF REPORT & PERIOD COVERED								
7. AUTHOR(s) S. T. Amimoto, J. S. Whittier, A. Whittaker, A. Chase, R. Hofland, Jr., and M. Bass		6. PERFORMING ORG. REPORT NUMBER TR-0084(4930-01)-2								
9. PERFORMING ORGANIZATION NAME AND ADDRESS The Aerospace Corporation El Segundo, Calif. 90245		8. CONTRACT OR GRANT NUMBER(s) F04701-83-C-0084								
11. CONTROLLING OFFICE NAME AND ADDRESS Space Division Los Angeles Air Force Station Los Angeles, Calif. 90009		10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS								
14. MONITORING AGENCY NAME & ADDRESS (if different from Controlling Office)		12. REPORT DATE 15 December 1983								
		13. NUMBER OF PAGES 20								
		15. SECURITY CLASS. (of this report) Unclassified								
		15a. DECLASSIFICATION/DOWNGRADING SCHEDULE								
16. DISTRIBUTION STATEMENT (of this Report) Approved for public release; distribution unlimited										
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report)										
18. SUPPLEMENTARY NOTES										
19. KEY WORDS (Continue on reverse side if necessary and identify by block number) <table border="0"> <tr> <td>Acid resistance</td> <td>DF-chain laser</td> </tr> <tr> <td>Adhesion strength</td> <td>Laser damage</td> </tr> <tr> <td>Carbyne (carbon) coatings</td> <td>Laser mirrors</td> </tr> <tr> <td>Coating absorption.</td> <td>Laser windows</td> </tr> </table>			Acid resistance	DF-chain laser	Adhesion strength	Laser damage	Carbyne (carbon) coatings	Laser mirrors	Coating absorption.	Laser windows
Acid resistance	DF-chain laser									
Adhesion strength	Laser damage									
Carbyne (carbon) coatings	Laser mirrors									
Coating absorption.	Laser windows									
20. ABSTRACT (Continue on reverse side if necessary and identify by block number) <p>→ Large-spot laser damage thresholds were measured for bowl-feed-polished CaF<sub>2</sub> and sapphire windows (bare and antireflection-coated) and for highly polished copper mirrors (bare and carbyne-coated) at DF chain-laser wavelengths (3.58-4.78 <math>\mu</math>m). The chain reaction between F<sub>2</sub> and D<sub>2</sub> was initiated by a magnetically confined electron beam, producing DF-laser outputs of 10 to 20 J in pulses of 0.6 to 0.9 <math>\mu</math>sec (FWHM) duration. Energy extracted from a transmission-coupled unstable resonator was focused by means of a</p>										

00 FORM  
(FACSIMILE)

1473

micrometers

microseconds

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE(When Data Entered)

19. KEY WORDS (Continued)

2.5 cm

20. ABSTRACT (Continued)

CaF<sub>2</sub> lens. A soft-aperture technique was employed to suppress effects of Fresnel diffraction so that uniform (top-hat) intensity profiles were obtained along the focusing beam. With this laser system, commercially available antireflection-coated CaF<sub>2</sub> and Al<sub>2</sub>O<sub>3</sub> samples were tested and found to have damage thresholds from 17 to 28 J/cm<sup>2</sup>. Significantly larger damage thresholds were observed for uncoated, polished samples of Al<sub>2</sub>O<sub>3</sub>, but damage resistance of uncoated polished CaF<sub>2</sub> was found to equal that of the best antireflection-coated CaF<sub>2</sub> samples. A highly polished copper mirror had the highest damage threshold of all the materials tested, i.e., 58 J/cm<sup>2</sup>. Carbyne films of diamond-like hardness, a type of carbon coating, were applied to polished copper mirrors and bowl-feed-polished CaF<sub>2</sub> surfaces. The carbyne coatings prepared in this work contained numerous carbon-bearing particles that were easily damaged (~10 J/cm<sup>2</sup>). However, regions of the irradiated carbyne film that were free of carbon particles withstood high laser fluences (25 J/cm<sup>2</sup>), indicating that improvements in carbyne film preparation would yield protective coatings with high damage resistance at DF wavelengths.

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE(When Data Entered)

# CONTENTS

I. INTRODUCTION.....	5
II. EXPERIMENTAL TECHNIQUE.....	7
III. RESULTS AND DISCUSSION.....	15
IV. CONCLUDING REMARKS.....	23
REFERENCES.....	25



Accession For	
NTIS GRA&I	<input checked="" type="checkbox"/>
DTIC TAB	<input type="checkbox"/>
Unannounced	<input type="checkbox"/>
Justification	
By	
Distribution/	
Availability Codes	
Dist	Avail and/or Special
A-1	

## FIGURES

1.	Laser Damage Apparatus Layout.....	8
2.	Calibrated Film Burns Showing Typical DF-Beam Spatial Fluence Distributions.....	9
3.	Diagram of Equipment Used to Apply Carbyne Films to Optical Components.....	12
4.	Carbyne Film on $\text{CaF}_2$ Irradiated at $23 \text{ J/cm}^2$ Showing Localized Damage at Carbon-Bearing Spots.....	17
5.	Micrograph of Bare Copper Mirror Showing Damage Sites at Exposure Fluence of $58 \text{ J/cm}^2$ for DF-Laser Pulse Duration of $0.5 \text{ } \mu\text{sec}$ (FWHM).....	18
6.	Micrographs of Copper Mirror Showing Substrate Regions Masked and Unmasked from Carbyne Vapor Stream.....	20



## I. INTRODUCTION

The performance of optical components can significantly affect the size, weight, and efficiency of high-power laser systems. Component failure in pulsed DF chain-reaction lasers presently imposes design limitations on laser gain length and aperture size. Unique challenges associated with component development for DF chain lasers are the laser's broadband spectral output of 3.58 to 4.78  $\mu\text{m}$ , the requirement for operation in the presence of a corrosive  $\text{F}_2/\text{DF}$  environment, and the need to withstand high impact loadings associated with combustion overpressures of from 6 to 8 atmospheres. Data on the failure of DF laser optical components are restricted primarily to small-energy pulses and, hence, to small spot sizes of the order of hundreds of micrometers.<sup>1,2</sup> The scaling of these results to large spot sizes of practical importance is known to be unreliable. Damage thresholds of 52  $\text{J}/\text{cm}^2$  were recently reported for  $\text{ThF}(\text{ZnS})_3$ -coated mirrors<sup>3</sup> using beam spots up to 1 cm in diameter.<sup>4</sup> Such excellent damage resistance has, however, been observed only in benign (air) environments. Similar coatings on internal mirrors and windows mounted in repetitively pulsed systems have repeatedly failed after several shots at incident fluences of several joules per square centimeter. Damage was by acid etching and by blowoff induced by laser irradiation.

Coating failure in pulsed chain-laser systems has prompted us to examine the suitability of carbyne coatings as hard protective films on pulsed DF-laser optical components. Chaoite, the carbyne form of interest, has been produced at The Aerospace Corporation by means of quench cooling of carbon gas.<sup>5,6</sup> Early studies of chaoite films revealed several interesting properties: (1) diamond-like hardness (greater than  $\text{B}_4\text{C}$ ), (2) chemical resistance to acids, bases, and organic solvents, (3) good adhesion to copper, platinum, glass, silicon, germanium, sapphire, and other materials, and (4) low absorption in the 2- to 40- $\mu\text{m}$  range. In view of these attractive properties, we undertook a brief program to apply carbyne films to candidate DF-laser window and mirror materials for evaluation with regard to adhesion strength, resistance to  $\text{HF}/\text{DF}$  attack, abrasion resistance, optical absorption, and laser

damage threshold. During the two-week study of laser damage resistance, we also measured damage thresholds for selected commercially available coatings applied to both transparent and reflective optics for purposes of comparison with chaoite film performance. The resulting data show that the best chaoite films have high damage resistance, hardness, and resistance to acid attack, and may be attractive for use on pulsed-DF-laser optical components if coatings can be developed that are free of carbon particles.

## II. EXPERIMENTAL TECHNIQUE

A diagram of the laser damage apparatus employed in the present study is given in Fig. 1. A magnetically confined electron beam was used to initiate the pulsed chain-reaction DF laser.<sup>7</sup> Mixtures containing 20% F<sub>2</sub>- 8% D<sub>2</sub> by volume were irradiated for periods of 0.1 to 1  $\mu$ sec at current densities of 20 A/cm<sup>2</sup> to accomplish laser initiation. Nominal laser energies of 10 to 20 J in 0.6 to 0.9  $\mu$ sec (FWHM) pulses were delivered at cavity pressures of 800 Torr. Energy was extracted from the gain medium by means of a transmission-coupled half-symmetric unstable resonator and then collimated using a CaF<sub>2</sub> lens of 8-m focal length. Laser windows were uncoated, 1.25-cm-thick CaF<sub>2</sub> crystals that were tilted with respect to the optical axis. A beam splitter at nearly normal incidence to the laser beam diverted about 6% of the total pulse energy into a 9-cm ballistic thermopile. Emission time history of the laser was monitored with a gold-doped germanium detector. In a previous study, the D<sub>2</sub>-F<sub>2</sub> laser spectral output was measured and found to consist of up to 69 lines operating between 3.58 and 4.78  $\mu$ m.<sup>8</sup>

The remaining energy in the pulsed DF-laser beam was focused by means of a CaF<sub>2</sub> lens of 45-cm focal length. The lens was translated along the direction of the beam to vary the fluence incident upon the optical test sample. A fluence range of 5 to 70 J/cm<sup>2</sup> could be encompassed by this technique. To obtain uniform beam spatial distributions along the focusing beam, an intra-cavity soft-aperture filter was employed (Fig. 1). This filter suppressed effects of Fresnel diffraction during focusing. The spatial distribution of the focusing beam was determined from burn patterns on calibrated witness film (Fig. 2). As illustrated in Fig. 2, a uniform (top-hat) spatial fluence distribution was obtained by use of the soft-aperture filter.

Laser damage measurements were carried out on a variety of transparent and reflective optics in the present work (Table I). Antireflection (ar)-coated windows of Al<sub>2</sub>O<sub>3</sub> and CaF<sub>2</sub> were obtained from Laser Power Optics (LPO) and CVI Laser Corporation. Coatings on the Al<sub>2</sub>O<sub>3</sub> samples were ZnS/ThF<sub>4</sub> and TiO<sub>2</sub>/Al<sub>2</sub>O<sub>3</sub>/MgF<sub>2</sub>; the ar coating on the CaF<sub>2</sub> sample was PbF<sub>2</sub>/ThF<sub>4</sub>. Details of

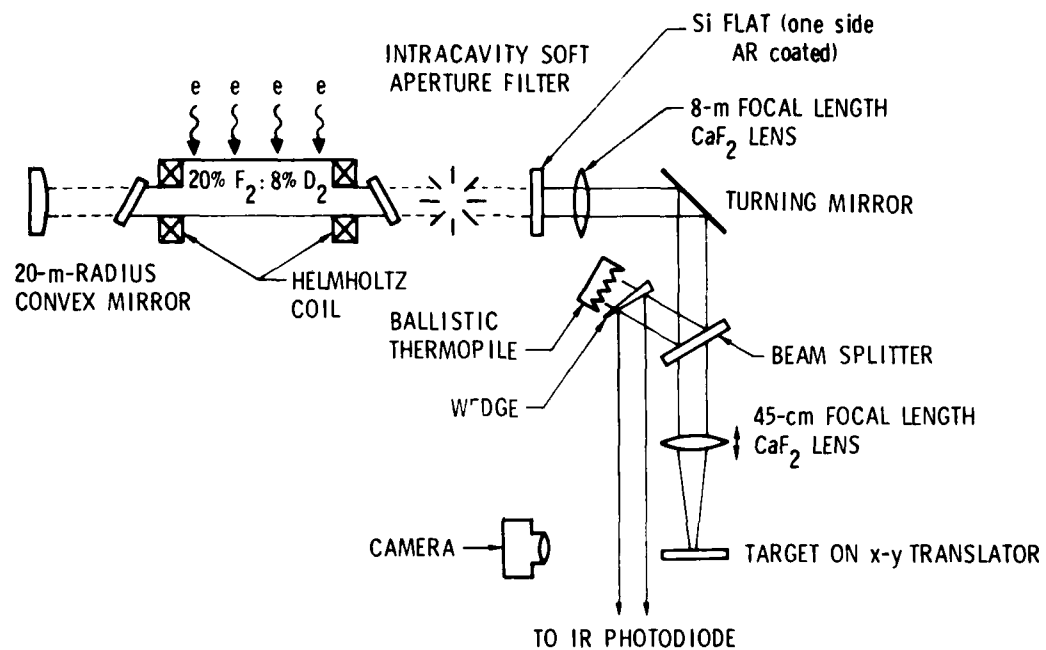


Fig. 1. Laser Damage Apparatus Layout

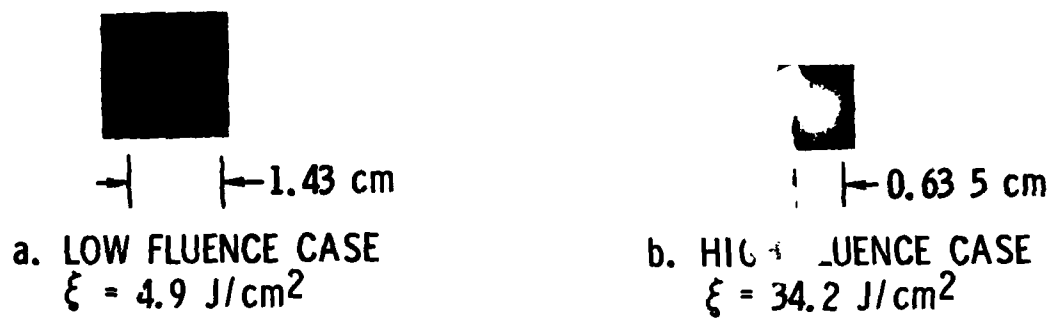


Fig. 2. Calibrated Film Burns Showing Typical DF-Beam Spatial Fluence Distributions

Table I. DF-Laser-Damage Summary

Substrate	Substrate Thickness, mm	Coating (Vendor)	Type of Damage	Fluence, J/cm <sup>2</sup>
Al <sub>2</sub> O <sub>3</sub>	6.25	None	Isolated spot on exit surface No entrance damage	25 ≤ 54
	3.12	TiO <sub>2</sub> /Al <sub>2</sub> O <sub>3</sub> /MgF <sub>2</sub> (CVI)	Isolated spot on exit surface coating Damage to entrance surface coating	28 38
	3.12	ZnS/ThF <sub>4</sub> (LPO)	Isolated damage on entrance and exit coating	17
CaF <sub>2</sub>	5	None	Exit surface and bulk damage at lens focus Entrance surface and bulk damage at lens focus	21 27
	5	PbF <sub>2</sub> /ThF <sub>4</sub> (LPO)	Exit coating damage Isolated spot on front	21 27
	5	Carbyne (Aerospace)	Sr <sub>11</sub> spot on entrance coating No visible damage	14 25
Cu	-	None (Spawr)	Slight surface damage	58
	-	Carbyne (Aerospace)	Severe coating failure	9.5

the coating designs are available from the vendors.<sup>9,10</sup> Bowl-feed-polished samples of  $\text{Al}_2\text{O}_3$  and  $\text{CaF}_2$  were also tested as a standard against which coated window performance could be assessed. Two mirrors were tested in our study: an uncoated laboratory-grade copper mirror (Spawr) and a carbyne-coated copper mirror. The performance of carbyne-coated  $\text{CaF}_2$  windows was also evaluated in our work.

Transparent carbon films for use in the present study were produced by quenching carbon gas on selected substrates, including transparent and reflective optical elements.<sup>11</sup> A schematic diagram of the equipment that was used to produce exploratory carbyne films is shown in Fig. 3. The sample to be coated and a piece of pure carbon were located near the center of the coating pressure vessel. The optical sample could be heated to  $180^\circ\text{C}$  and was discharge cleaned prior to coating application. The pure carbon source was positioned in a carbon mandrel that was held in the chuck of a spinner mounting. The spinning carbon target was heated by means of a focused 1.5 kW  $\text{CO}_2$  laser beam that entered the chamber through a NaCl window. During laser irradiation, the hot carbon rod was surrounded by a cloud of carbon gas that impinged on the surface of the spinning optical sample. The temperature of the solid carbon source was measured by an optical pyrometer of fast response time ( $\sim 0.1$  sec). The pyrometer was located, as shown in Fig. 3, with its line of sight at  $30^\circ$  to the laser beam axis. Careful temperature control of the carbon source was crucial for production of the desired carbyne film. Chamber pressure was measured by a Baratron gauge. Instability in the output of the  $\text{CO}_2$  laser resulted in difficulties in the preparation of uniform, reproducible coatings. In view of the unsophisticated nature of the coating apparatus, it is believed that improvement in the quality and reproducibility of carbyne coatings should be readily achievable.

Fifty laser shots were performed on the 10 samples that were available for our damage study. Approximately one full day of testing was required to determine the damage threshold of an optical component. During testing, the sample was rotated after each exposure so that a fresh area was irradiated. The large spot sizes used in our study implied that only a few exposures could

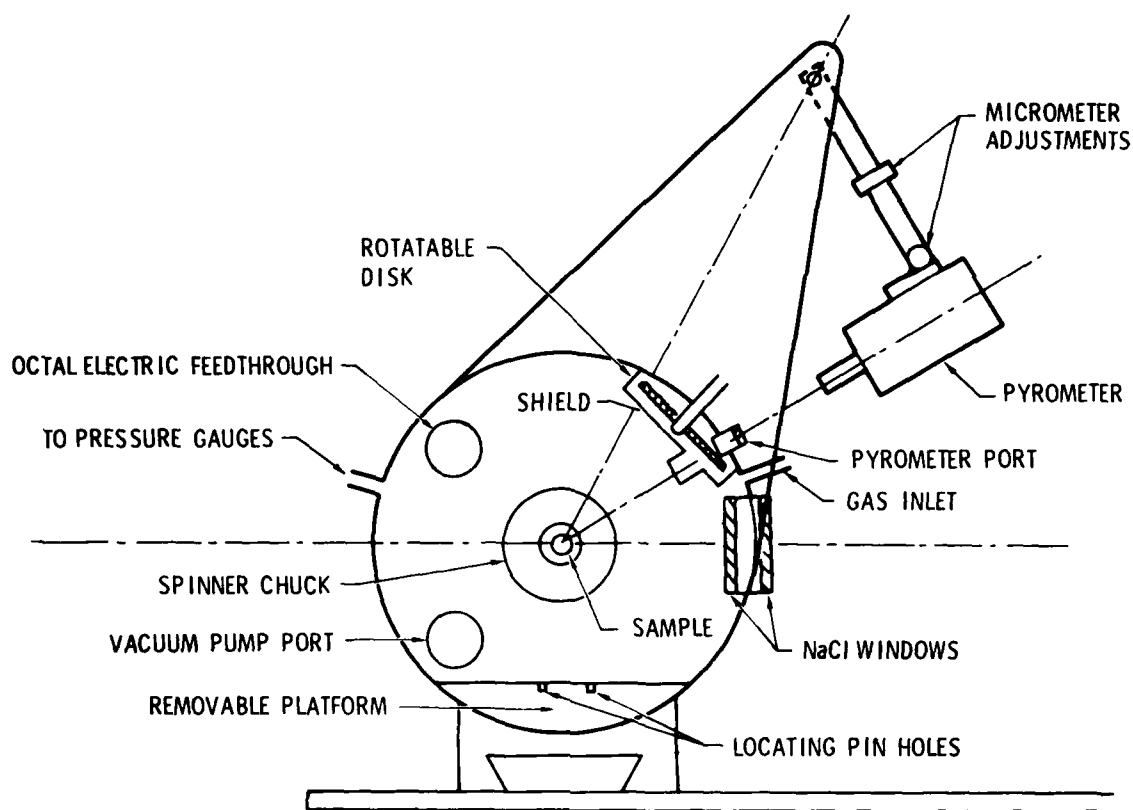


Fig. 3. Diagram of Equipment Used to Apply Carbyne Films to Optical Components



be performed on each optical sample. Exposure began below  $10 \text{ J/cm}^2$  and was increased in steps of 50 to 100% until small-scale and, finally, catastrophic damage was observed. Component damage was determined by post-irradiation inspection with a 10X microscope, using strong illumination.

### III. RESULTS AND DISCUSSION

Selected properties of carbyne films were briefly examined before measurement of laser damage resistance. Absorption coefficients in the infrared were found to increase from approximately  $2 \times 10^2 \text{ cm}^{-1}$  at  $10 \mu\text{m}$  to approximately  $2 \times 10^3 \text{ cm}^{-1}$  at  $1.0 \mu\text{m}$ . Index of refraction averaged about 2.1; one film, however, gave a rather low index value of 1.7. Absorption peaks at 3.1, 5.9, 6.2, 7.3, and  $8.5 \mu\text{m}$  strongly indicated that the present carbyne films were contaminated with pump oil. Contamination was possible because no precautions were taken to prevent backstreaming of pump oil during the carbyne coating process. Because pump oil would be expected to increase absorption in the 3.6 to  $4.8 \mu\text{m}$  bandpass, its presence probably would have reduced the DF damage resistance of the carbyne-coated samples that were tested in this work.

An ion-microprobe mass analyzer (IMMA) was used to obtain unequivocal identification of carbyne as the film that was produced. Although two different carbon negative-ion spectra have been observed using the IMMA, carbon films used in this study gave spectra that stopped at  $\text{C}_4$ . Unfortunately, the IMMA could not identify the particular carbyne form that was produced in our work.

Abrasion tests were also performed on the carbyne films. The results revealed that the films were more abrasion resistant than their substrates (Ni, Al, Cu,  $\text{CaF}_2$ ); they were determined, however, to be less resistant than fused quartz. Adhesion strength of the carbyne films varied over the range from 57 to  $652 \text{ kg/cm}^2$ . The diffusion of vacuum pump oil onto the substrates could have accounted for these low values as well as the wide variation in adhesion strength.

Chemical resistance of the carbyne films was evaluated using concentrated HF acid. The films were unaffected by the acid. Penetration of the films by way of imperfections was observed, however. One continuous film was tested that was not penetrated during the time required for the droplet of acid to evaporate (about 30 minutes).

The laser damage threshold measurements performed during the present study are summarized in Table I. Included in the table are component substrate material, substrate thickness, component coating (when present), coating vendor, type of damage observed, and incident laser fluence at which damage was first detected. For the sapphire substrate case, both the ar coating of  $\text{TiO}_2/\text{Al}_2\text{O}_3/\text{MgF}_2$  and the bare substrate were found to exhibit high surface-damage thresholds, i.e., 28 to  $54 \text{ J/cm}^2$ . As anticipated, the exit surfaces were damaged at thresholds well below those of the entrance surface. The damage threshold of the  $\text{ZnS}/\text{ThF}_4$  ar coating ( $17 \text{ J/cm}^2$ ) was found to be the lowest of all the sapphire samples tested.

Damage tests on  $\text{CaF}_2$  substrates showed that commercially available ar coatings have damage resistance equal to that of the uncoated  $\text{CaF}_2$  surface (Table I). The carbyne coatings on  $\text{CaF}_2$  exhibited damage thresholds that varied over a wide range of fluences ( $14\text{--}25 \text{ J/cm}^2$ ). The best carbyne film was observed to have a damage resistance equal to that of the polished  $\text{CaF}_2$  surface. On a microscopic scale, the coatings exhibited sample-to-sample variations as well as variations across a given sample. The damage levels could not be correlated with any particular deposition procedure. Figure 4 is a micrograph showing damage to a carbyne film deposited on  $\text{CaF}_2$  and exposed at  $23 \text{ J/cm}^2$  incident laser fluence. The damage is seen to have occurred preferentially at carbon-bearing spots in the carbyne film. It is likely that these carbon particles absorbed the laser radiation preferentially, but the regions remote from the damage spots were observed to be coated with a carbyne film that did not degrade at laser fluence levels at which the better commercially available films did degrade. Carbyne coatings of very high damage resistance should be achievable, therefore, if the coatings can be made free of particulate carbon matter.

A limited number of damage tests were performed on reflective optical components (Table I). An oxygen-free, high-conductivity (OFHC) copper mirror of "laboratory grade" surface finish was measured to have the highest damage threshold of all the component samples that were tested. Figure 5 is a micrograph of the laser damage sites on this polished mirror. The degree of damage is seen to be quite small, considering the high exposure fluence:



Fig. 4. Carbyne Film on  $\text{CaF}_2$  Irradiated at  $23 \text{ J/cm}^2$   
Showing Localized Damage at Carbon-Bearing (black)  
Spots. Magnification 200X



Fig. 5. Micrograph of Bare Copper (Spawr) Mirror Showing Damage Sites at Exposure Fluence of  $58 \text{ J/cm}^2$  and DF-Laser Pulse Duration of  $0.5 \text{ } \mu\text{sec}$  (FWHM). Magnification 37.5X

58 J/cm<sup>2</sup>. The carbyne-coated copper samples consistently failed at low fluences of 9.5 J/cm<sup>2</sup> or less. During the coating of copper samples, one half of each substrate was masked from the hot carbon source. Figure 6a gives clear evidence that particulate carbon contamination occurred on the half of the copper mirror that was considered to have been shielded from the heated carbon source. A method for interception of these carbon particulates must be devised for future coating studies. The irradiation of the carbyne vapor stream by a high-power laser beam would be one technique for interception of particulate material before its impingement on a substrate surface. Figure 6b is a micrograph showing laser damage to a carbyne film deposited on half of a polished copper substrate. Preferential damage at carbon-bearing sites is seen to be the failure mechanism, as was observed for the case of carbyne coatings on CaF<sub>2</sub> samples. We speculate that carbyne films on reflective optics would possess high damage resistance in the absence of these particulate carbon sites.

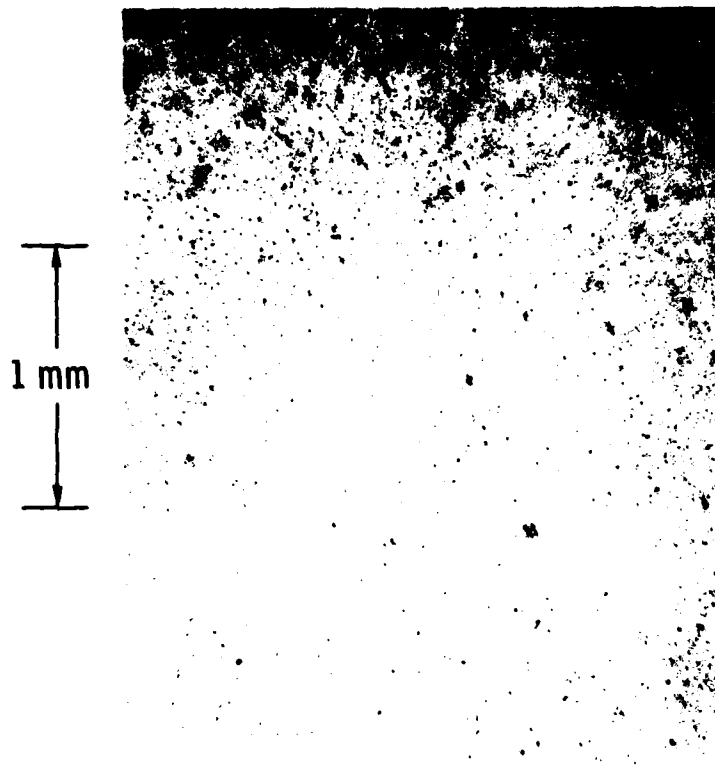


Fig. 6. Micrographs of Copper Mirror Showing Substrate Regions Masked and Unmasked from Carbyne Vapor Stream. Localized laser damage to carbyne coating occurs at carbon particulate sites. a. Masked Mirror Segment. No exposure to laser radiation. (Original magnification 37.5X) Carbon particulates have contaminated this copper substrate despite attempts to shield mirror segment from the carbyne effluent.

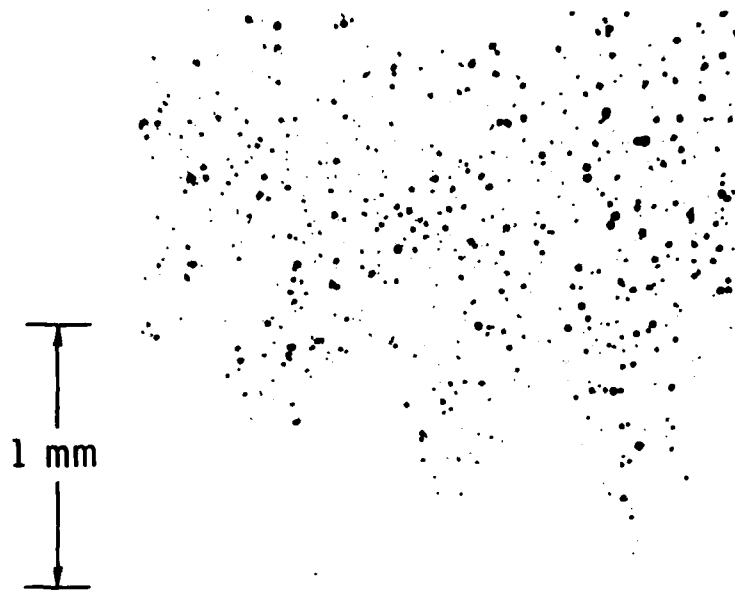


Fig. 6b. Unmasked Mirror Segment. Lower half not exposed to laser radiation. Upper half exposed to  $9.5 \text{ J/cm}^2$ . (Original magnification 37.5X)



#### IV. CONCLUDING REMARKS

Large-spot DF-laser damage thresholds have been investigated for bare substrates, commercially coated components, and hard carbyne films deposited on transparent and reflective optics. Damage resistance measurements on candidate DF-laser window components have shown that high-quality antireflection coatings are presently available for  $\text{CaF}_2$  and  $\text{Al}_2\text{O}_3$  substrates. Damage resistance of OFHC polished-copper mirrors has been excellent at DF chain-laser wavelengths. Unfortunately, the ability of these components to withstand high radiation fluxes in the presence of the hot corrosive gas flows of repetitively-pulsed systems is limited. Because of this limitation, we examined Aerospace carbyne coatings as candidate protective films for pulsed DF optical components. In our study, the best carbyne coatings survived high laser fluences of  $25 \text{ J/cm}^2$ . However, carbyne films as currently prepared for laser-damage evaluation generally showed numerous particulate carbon sites that were easily damaged. Areas free of these defects possessed high laser damage resistance. We recommend, therefore, that improvements in carbyne-coating preparation be pursued as a step toward the ultimate development of practical, damage-resistant films for use both in repetitively-pulsed DF laser devices and in other optical systems where corrosion resistance and high laser-damage resistance are essential characteristics.

## REFERENCES

1. E. W. Van Stryland, M. Bass, M. J. Soileau, and C. C. Tang, Pulsed HF/DF Laser Damage in Window Materials, NBS Special Pub. 509, 1977.
2. T. M. Donovan, J. O. Porteus, S. C. Seitel, P. Kraatz, Multithreshold HF/DF Pulsed Laser Damage Measurements on Evaporated and Sputtered Silicon Films, NBS Special Pub. 620, 305 (1980).
3. P. Kraatz, Optical Coatings for 2-6 Microns, Northrop Resh. and Techn. Ctr. Rpt. NRTC78-13R, April 1978.
4. D. B. Nichols, R. B. Hall and R. A. House, Large Spot DF Laser Damage of Dielectric-Enhanced Mirrors, to be published.
5. A. G. Whittaker, Science 200, 763 (1978).
6. A. G. Whittaker, Carbon 17, 21 (1979).
7. S. T. Amimoto, J. S. Whittier, M. L. Lundquist, F. G. Ronkowski, P. J. Ortwerth, R. Hofland, Appl. Phys. Lett. 40, 20 (1982).
8. S. T. Amimoto, DF Chain-Laser Spectra, TR-0081(6404)-1, The Aerospace Corp. (Oct. 1980).
9. G. Hahn, CVI Laser, private communication.
10. G. H. Sherman, Laser Power Optics, private communication.
11. A. G. Whittaker, P. L. Kintner, L. S. Nelson, N. Richardson, Rev. Sci. Instrum. 48, 632 (1977).

#### LABORATORY OPERATIONS

The Laboratory Operations of The Aerospace Corporation is conducting experimental and theoretical investigations necessary for the evaluation and application of scientific advances to new military space systems. Versatility and flexibility have been developed to a high degree by the laboratory personnel in dealing with the many problems encountered in the nation's rapidly developing space systems. Expertise in the latest scientific developments is vital to the accomplishment of tasks related to these problems. The laboratories that contribute to this research are:

Aerophysics Laboratory: Launch vehicle and reentry aerodynamics and heat transfer, propulsion chemistry and fluid mechanics, structural mechanics, flight dynamics, high-temperature thermomechanics, gas kinetics and radiation; research in environmental chemistry and contamination; cw and pulsed chemical laser development including chemical kinetics, spectroscopy, optical resonators and beam pointing; atmospheric propagation, laser effects and countermeasures.

Chemistry and Physics Laboratory: Atmospheric chemical reactions, atmospheric optics, light scattering, state-specific chemical reactions and radiation transport in rocket plumes, applied laser spectroscopy, laser chemistry, battery electrochemistry, space vacuum and radiation effects on materials, lubrication and surface phenomena, thermionic emission, photosensitive materials and detectors, atomic frequency standards, and bioenvironmental research and monitoring.

Electronics Research Laboratory: Microelectronics, GaAs low-noise and power devices, semiconductor lasers, electromagnetic and optical propagation phenomena, quantum electronics, laser communications, lidar, and electro-optics; communication sciences, applied electronics, semiconductor crystal and device physics, radiometric imaging; millimeter-wave and microwave technology.

Information Sciences Research Office: Program verification, program translation, performance-sensitive system design, distributed architectures for spaceborne computers, fault-tolerant computer systems, artificial intelligence, and microelectronics applications.

Materials Sciences Laboratory: Development of new materials: metal matrix composites, polymers, and new forms of carbon; component failure analysis and reliability; fracture mechanics and stress corrosion; evaluation of materials in space environments; materials performance in space transportation systems; analysis of systems vulnerability and survivability in enemy-induced environments.

Space Sciences Laboratory: Atmospheric and ionospheric physics, radiation from the atmosphere, density and composition of the upper atmosphere, aurorae and airglow; magnetospheric physics, cosmic rays, generation and propagation of plasma waves in the magnetosphere; solar physics, infrared astronomy; the effects of nuclear explosions, magnetic storms, and solar activity on the earth's atmosphere, ionosphere, and magnetosphere; the effects of optical, electromagnetic, and particulate radiations in space on space systems.